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Stability analysis for a wind turbine blade

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Problem Definition:

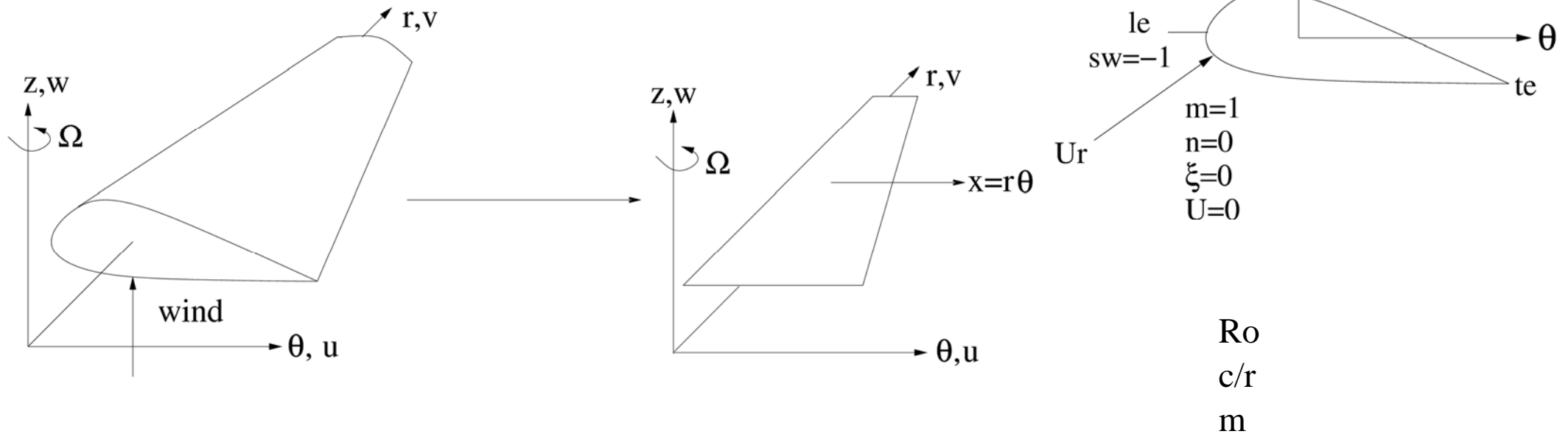
- Transition prediction for improving design of wind turbine airfoils.
- Demand a method not expensive in terms of computing time.
- Database approach easy to integrate in any numerical code.

Velocity profiles with rotational effects ---> Stability analysis

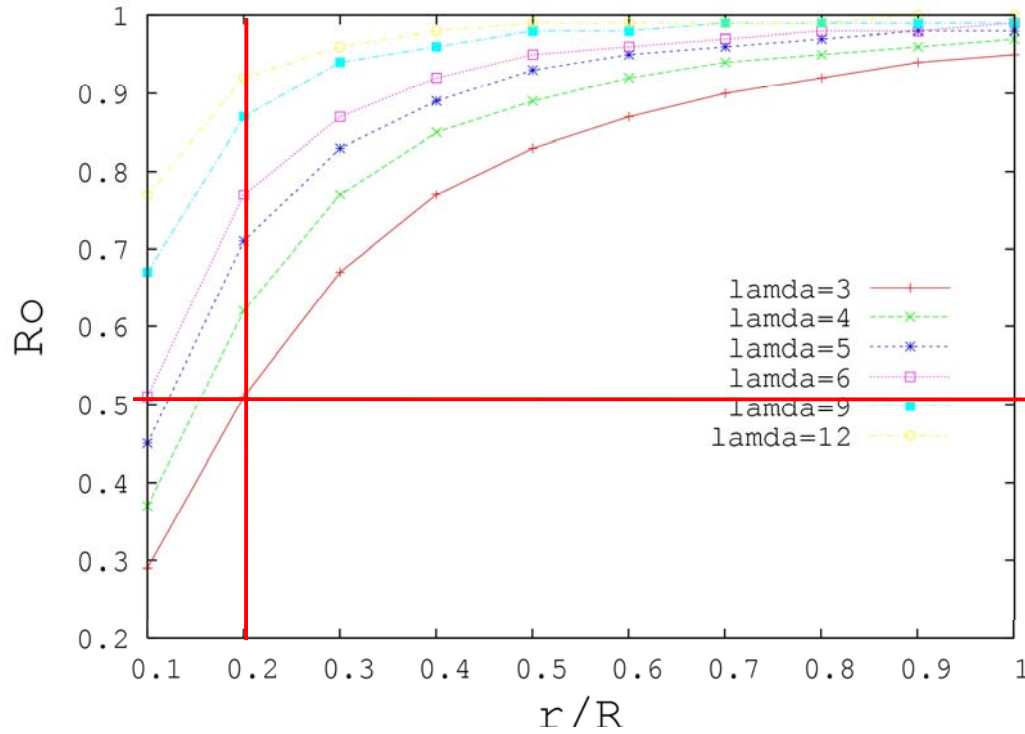


Implementation on EllipSys3D code <--- Amplification Factor

Boundary layer Velocity profiles Database:



- The velocity profiles have been mapped and stored.
- Unique relation exist if the dimensionless pressure gradients m and n , with the Rossby number are specified. (True for a rotating flat plate). In a similar way another parameters can be selected: dimensionless wall shear stress in both directions and the Rossby number of the shape factors.
- Relations among the parameters have been obtained.



$$Ro = \frac{\lambda}{\sqrt{\lambda^2 + (1-a)^2}}$$

$$r/R = \frac{\lambda}{\lambda_g}$$

$$f''(0) \equiv \frac{1}{2} C_{f,u} \sqrt{Re_x}$$

$$g''(0) \equiv \frac{1}{2} C_{f,v} \sqrt{Re_x}$$

Parameter Range:

$$\lambda_g = [3-12]$$

$$\lambda = \left[\frac{\lambda_g}{ns} - \lambda_g \right]$$

$$c/r = [0.50-0]$$

$$Ro = [0.50-0.95]$$

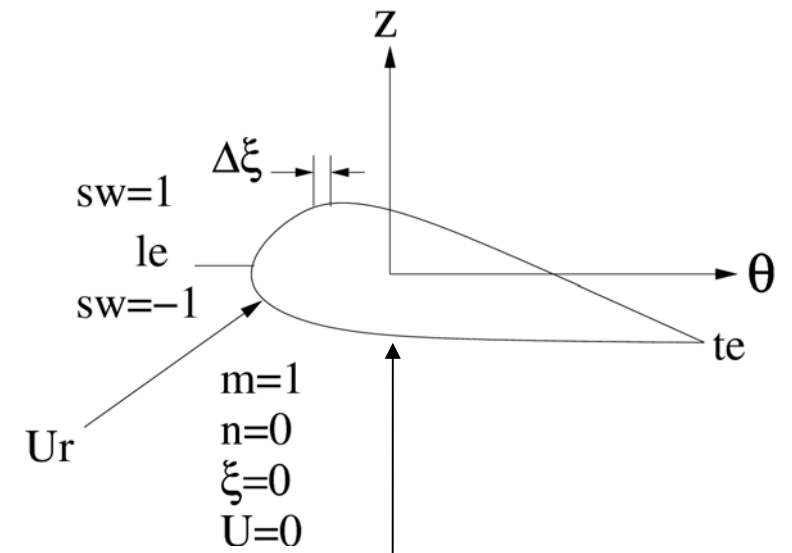
Pressure

Gradient:

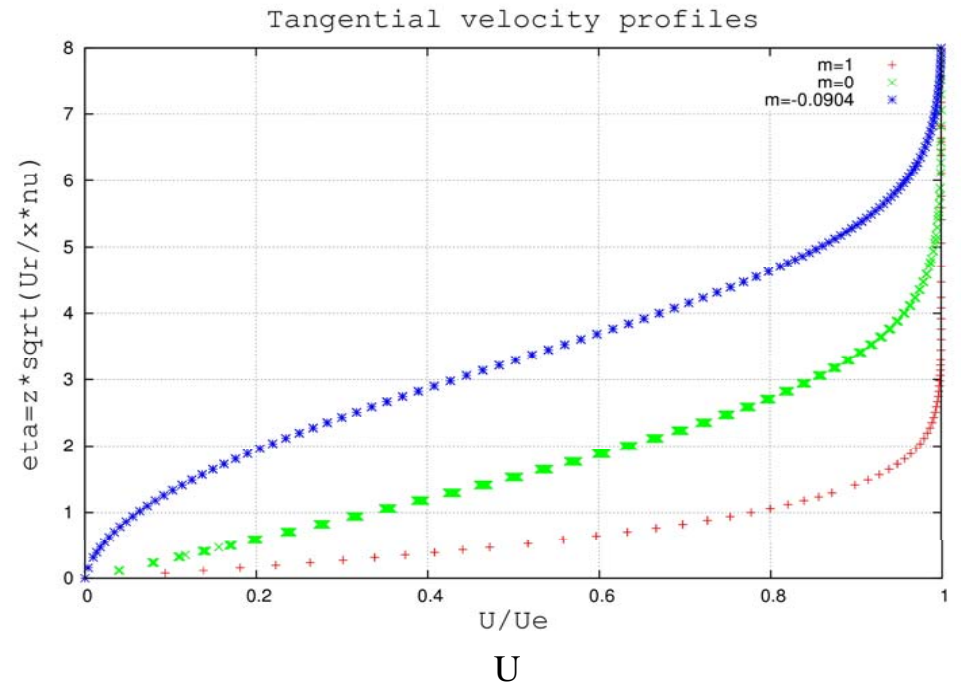
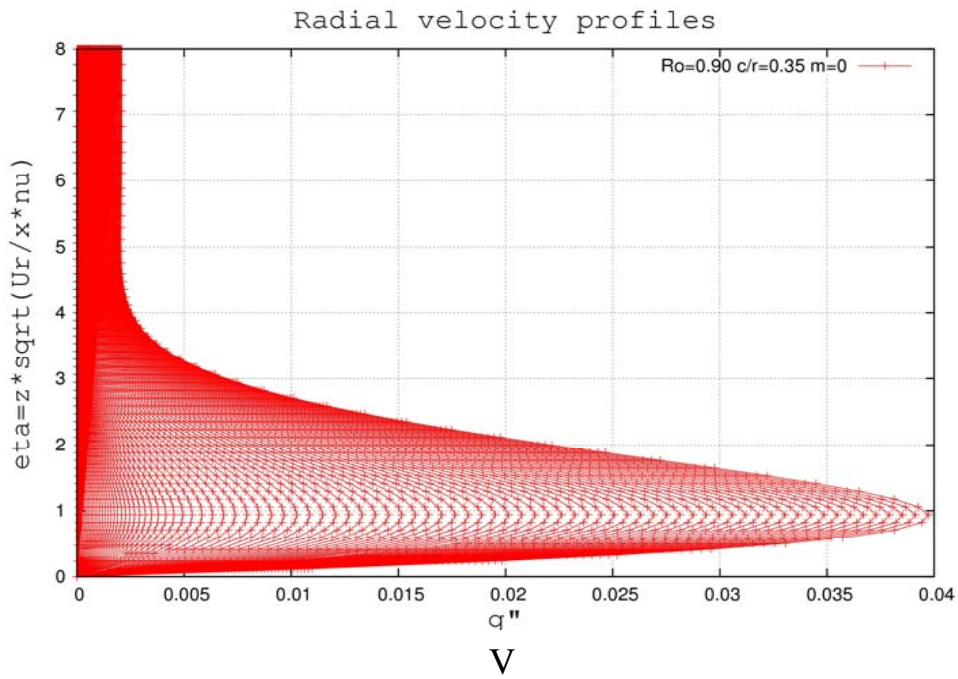
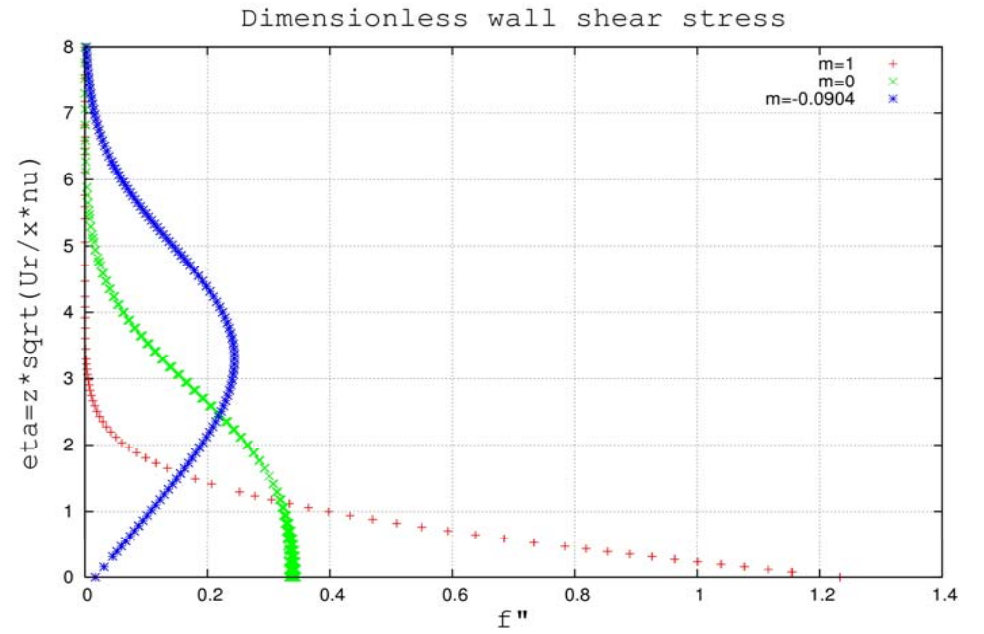
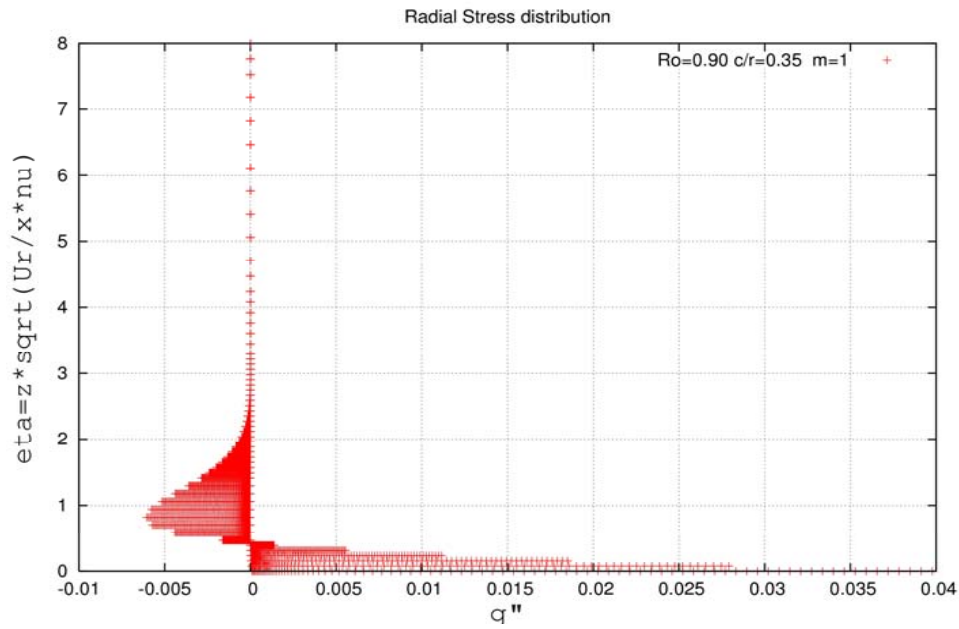
$$\alpha = [0-20^\circ]$$

$$m = \frac{\xi}{U} \frac{dU}{d\xi} \frac{1}{S_t}$$

$$n = \frac{c}{r} Ro^2 \xi$$



Input for the stability Analysis:



Orr-Sommerfeld equation:

- Based on parallel flow assumption.
- Linear stability is used in the derivation the equation.
- Disturbance shape equation:

$$r' = r(y)e^{[i(\alpha x + \beta z - \omega t)]} \quad \alpha = \frac{2\pi}{\lambda_x} \quad \beta = \frac{2\pi}{\lambda_z} \quad \omega = \frac{c}{\alpha}$$

- Spatial vs temporal approach.

$$\varphi'''' - 2\alpha^2 \varphi'' + \alpha^4 \varphi - iR[(\alpha U - \omega)(\varphi'' - \alpha^2 \varphi) - \alpha U'' \varphi] = 0$$

$$\varphi'''' - 2(\alpha^2 + \beta^2) \varphi'' + (\alpha^2 + \beta^2)^2 \varphi - iR[(\alpha U + \beta V - \omega)(\varphi'' - (\alpha^2 + \beta^2) \varphi) + (\alpha U'' + \beta V'') \varphi] = 0$$

Boundary Conditions:

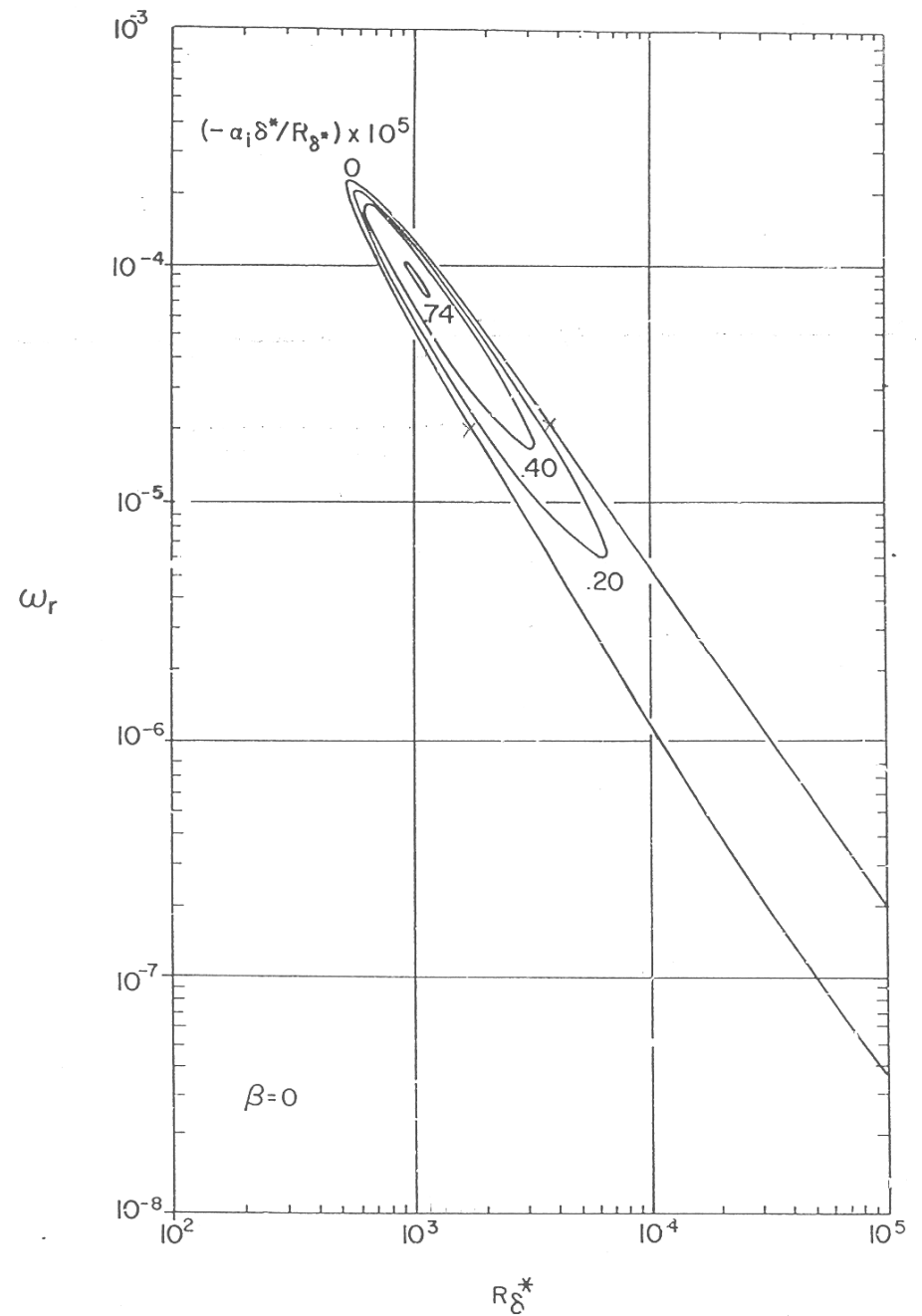
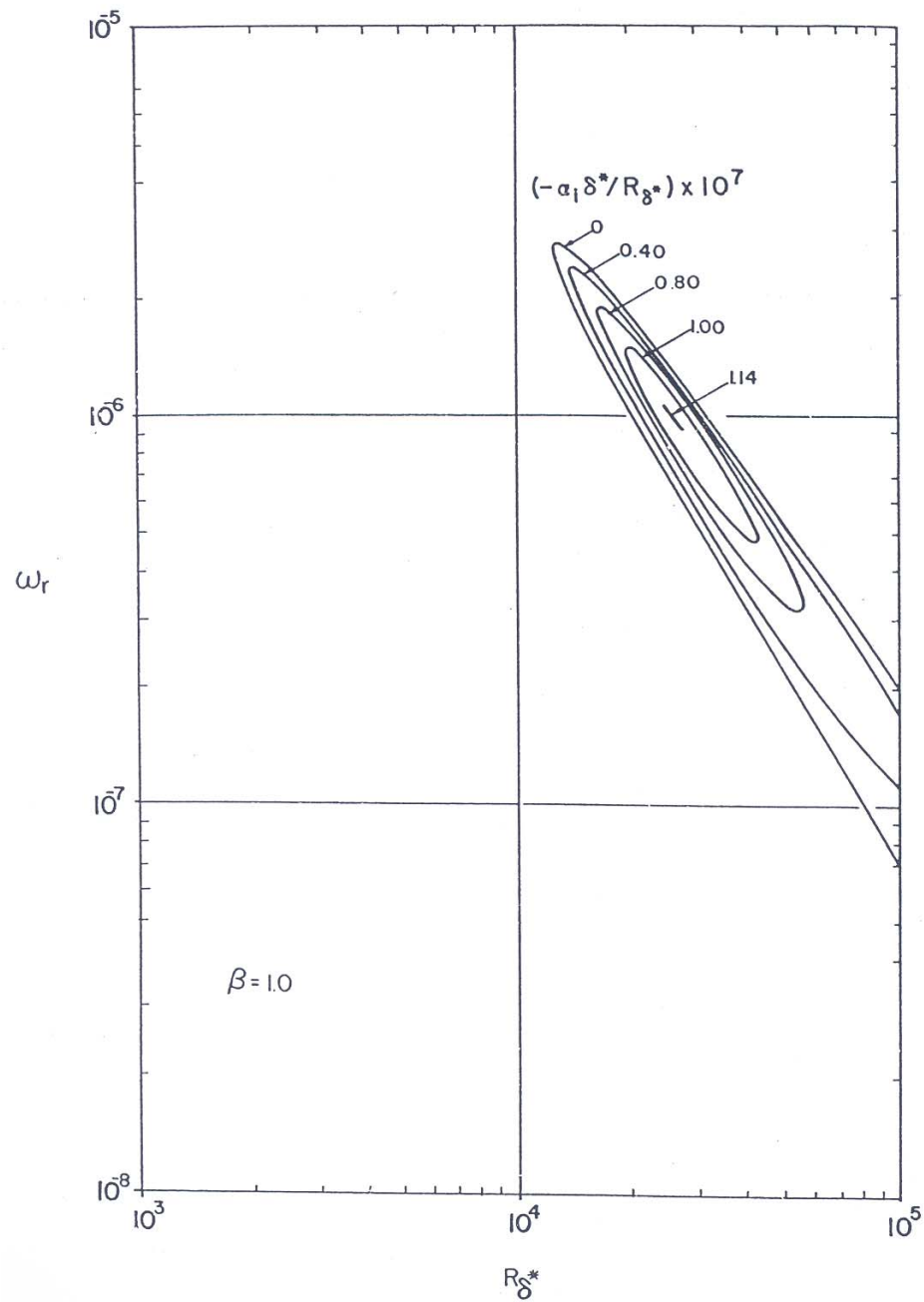
$$y = 0, \varphi = 0, \varphi' = 0$$

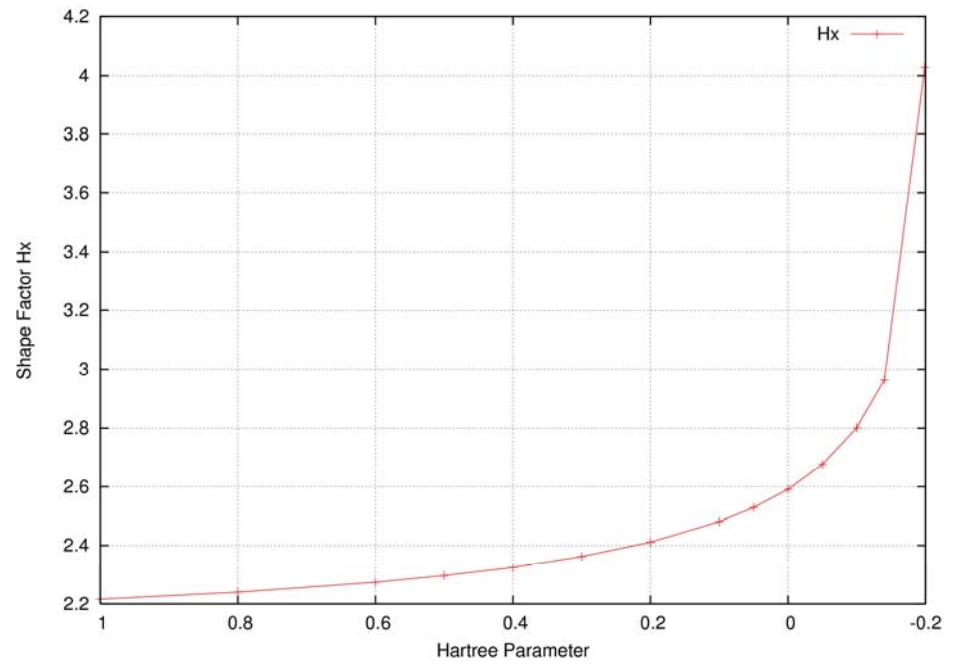
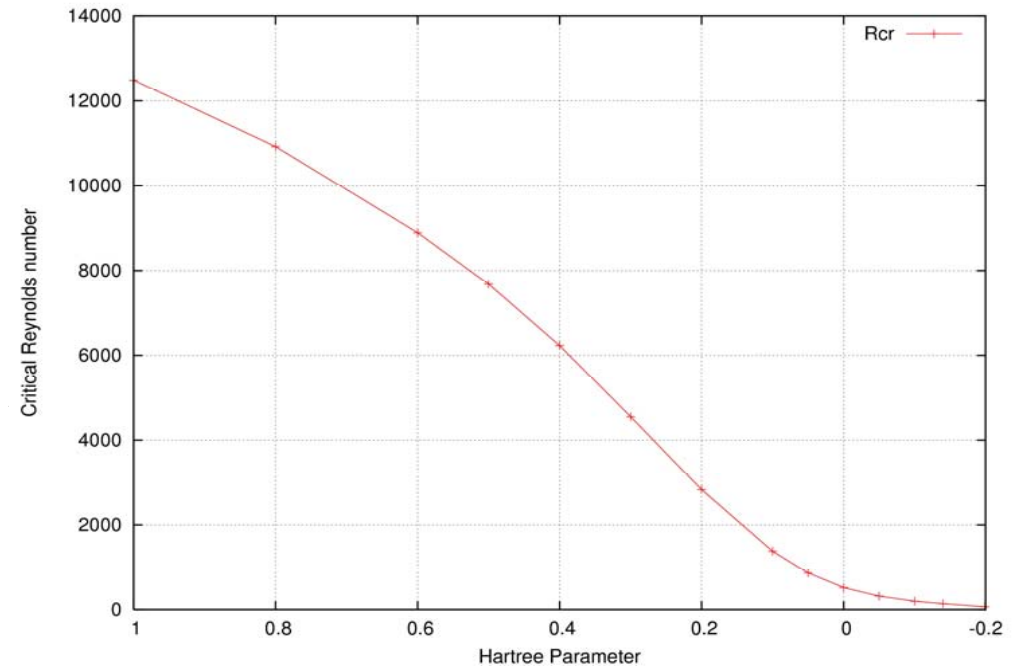
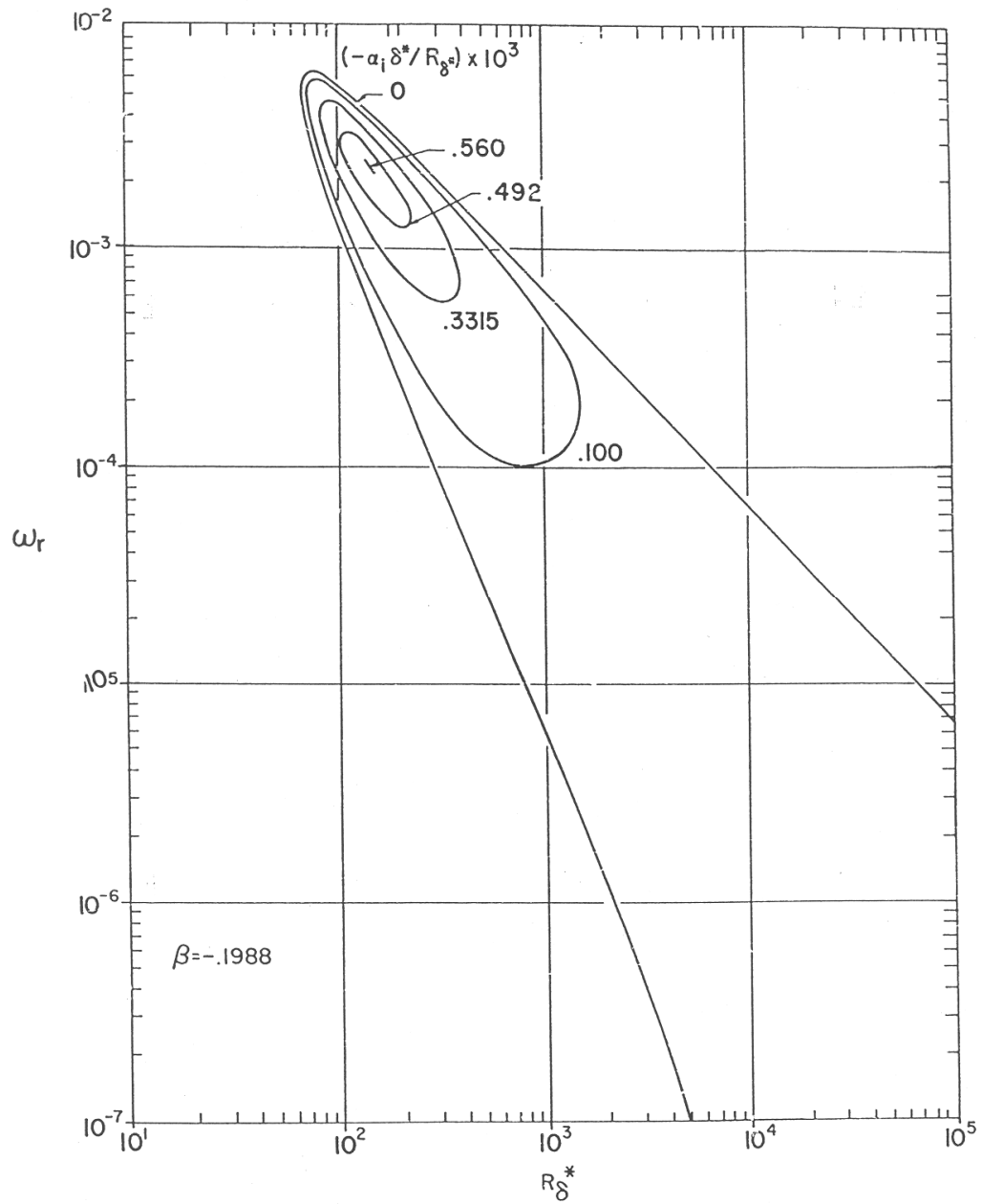
Eigenvalue Problem

$$F(\alpha, \omega, R) = 0 \quad F(\alpha, \beta, \omega, R) = 0$$



Tangential flow Database :





What about the cross flow, how to handle the problem ?

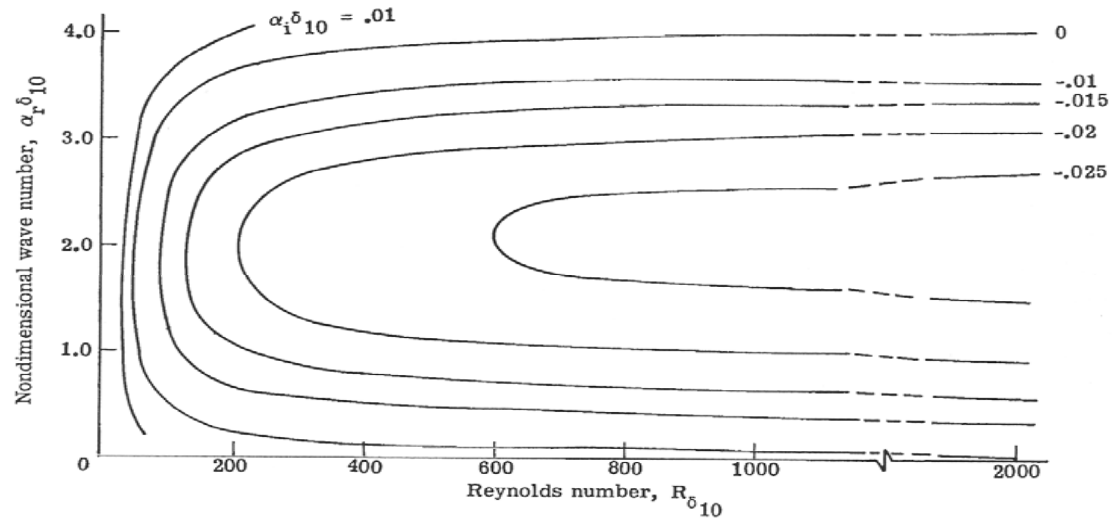
Proposed approach to solve for the Cross flow Database:

- One very important conclusion from the analysis of the velocity profiles is that the tangential flow is not affected by the cross flow velocity profile. Therefore it is possible to consider the stability of the two velocity profiles independently.

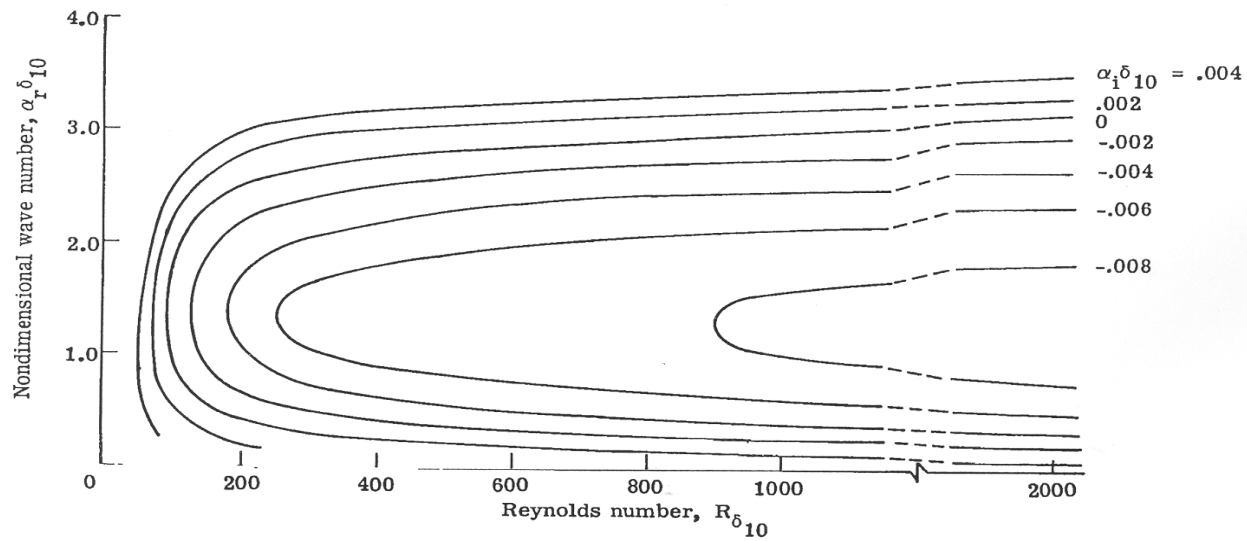
- It is important to notice that the method used to solve the eigenvalue problem is very closely related to the method used to define the transition criteria.

- From *Annu. Rev. Fluid. Mech.* 1996. 28, H Reed, W.S. Saric and D. Arnal, for the case of subsonic flows, recommend to separate or uncouple the problem of the cross flow instability from the TS instability, since the mechanisms are different.

Stability Diagram for the cross flow velocity profile:



(c) $H_c = 0.4299$ and $w_M/U_{e,t} = -0.04624$.



(d) $H_c = 0.2444$ and $w_M/U_{e,t} = 0.01678$.

Figure 11.- Continued.

The approach requires the following:

- Reformulation of the Orr-Sommerfeld equation and the parameters in the stability diagrams, (Reynolds number) and disturbance shape and boundary conditions.
- Two stability modes are considered: Traveling Cross flow waves and stationary cross flow instability.
- Alternative formulation of the problem for instance the envelope method, to verify the solution. A kind of Orr-Sommerfeld equation including rotational effects have been derived. (Still need to check the boundary conditions for solution)



Thank you :)