

Qualitative and Quantitative Characterization of a Jet and Vortex Actuator



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Outline

- Motivation
- Construction of a JaVA Prototype
- Experimental Setup and Governing Parameters
- JaVA Visualization Results
- Extraction of Optical Flow Fields
- Conclusions & Outlook

Known & Open Issues

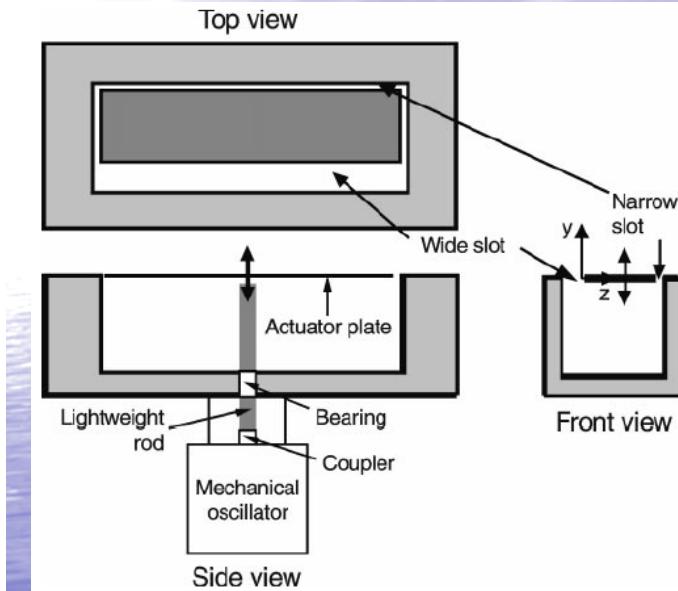
- Synthetic Jets
 - Operate over a range of amplitudes and frequencies
 - Don't need any external fluid (zero-net-mass flux system)
 - Add only negligible drag when the system is not actuated
- JaVA (Lachowicz *et al.*, Exp. Fluids, 1999)
 - All of the above
 - + Very wide-spread flow-fields in still air (e.g. wall and free jets of different orientation, vortex flow, direct generation of vortices)
 - How does it work?
 - What potential applications ?

References

- [1] Lachowicz, J. T., Yao, C., and Wlezien, R. W., 1999: *Flow Field Characterization of a Jet and Vortex Actuator*, Experiments in Fluids, Vol. 27, pp. 12-20
- [2] Joslin, R.D., Lachowicz, J. T., and Yao, C., 1998: *DNS of Flow Induced by a Multi-Flow*, ASME paper Number FEDSM98-5302, ASME Fluid Engineering Meeting
- [3] Koumoutsakos, P., 1995: *Simulations of Vortex Generators*, Center for Turbulence Research, Annual Research Briefs, pp. 233-240
- [4] Saddoughi, S.G., Koumoutsakos, P., Bradshaw, P., Mansour N.N., 1998: *Investigation of on demand vortex generators*, Center for Turbulence Research Manuscript No. 171, Stanford University
- [5] Jacobson, S.C. and Reynolds, W.C., 1998: *Active control of stream wise vortices and streaks in boundary layers*, Journal of Fluid Mechanics **360**: 179-994

Basic Setup & Parameters

Lachowicz *et al.*, Exp. Fluids, 1999



Nondimensional governing parameters:

Actuator amplitude a [mm]

- Actuator width b [mm]
- $\pi_1 = Ub/U$
- $\pi_2 = f/Re = 2\pi afb/U$
- Actuator frequency f [Hz]
- Characteristic velocity U [mm/s]
- $\pi_3 = U/bf$
- Wide slot spacing w_w [mm]
- $\pi_4 = w_n/w_w$
- Narrow slot spacing w_n [mm]
- Kinematic viscosity $\nu = S_a U = 2\pi a^2 b/U$ [mm²/s]
- $\pi_5 = g_n = w_n/b$

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Results of Lachowicz *et al.*, Exp. Fluids, 1999

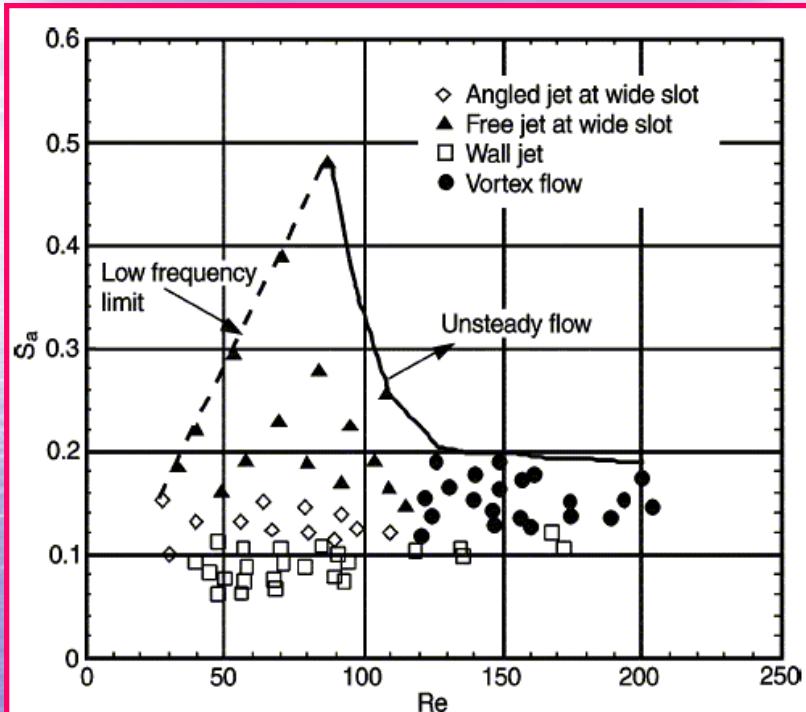


Fig. 9. Flow Regimes at $g_w = 0.099$

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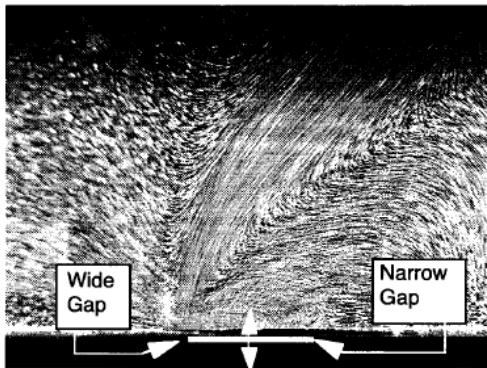


Figure 2. Angled jet at wide gap: $f=70$ Hz, $g_r=3$, $S_a=0.13$, $Re=56$

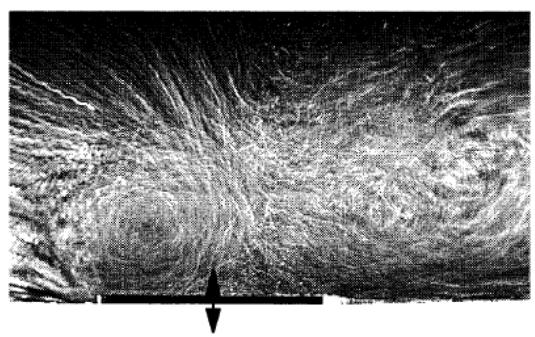


Figure 5. Vortex flow: $f=190$ Hz, $g_r=3$, $S_a=0.13$, $Re=146$

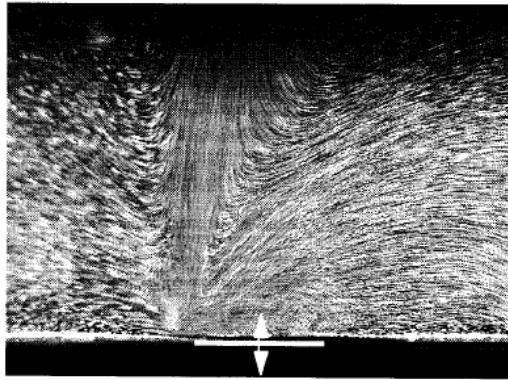


Figure 3. Jet at wide gap: $f=70$ Hz, $g_r=3$, $S_a=0.19$, $Re=80$

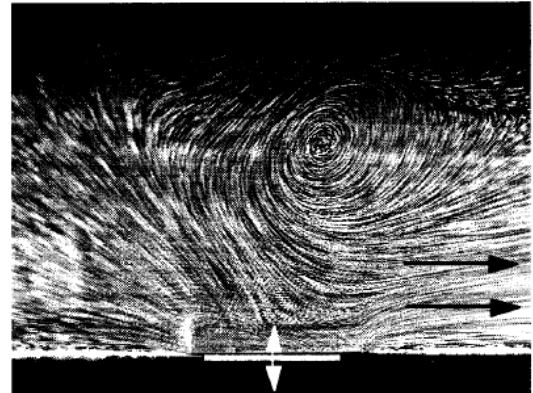
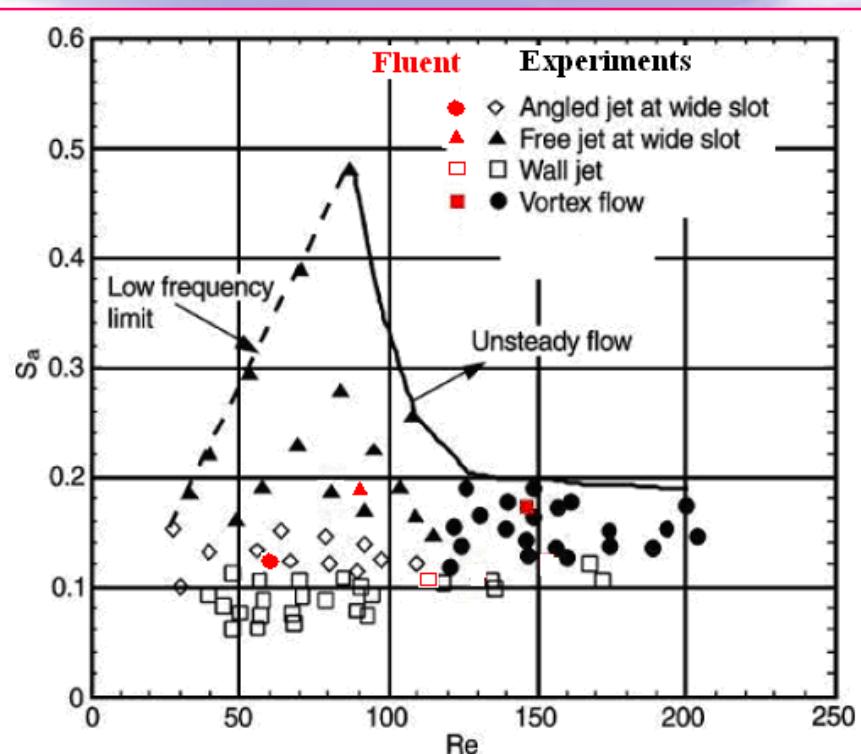


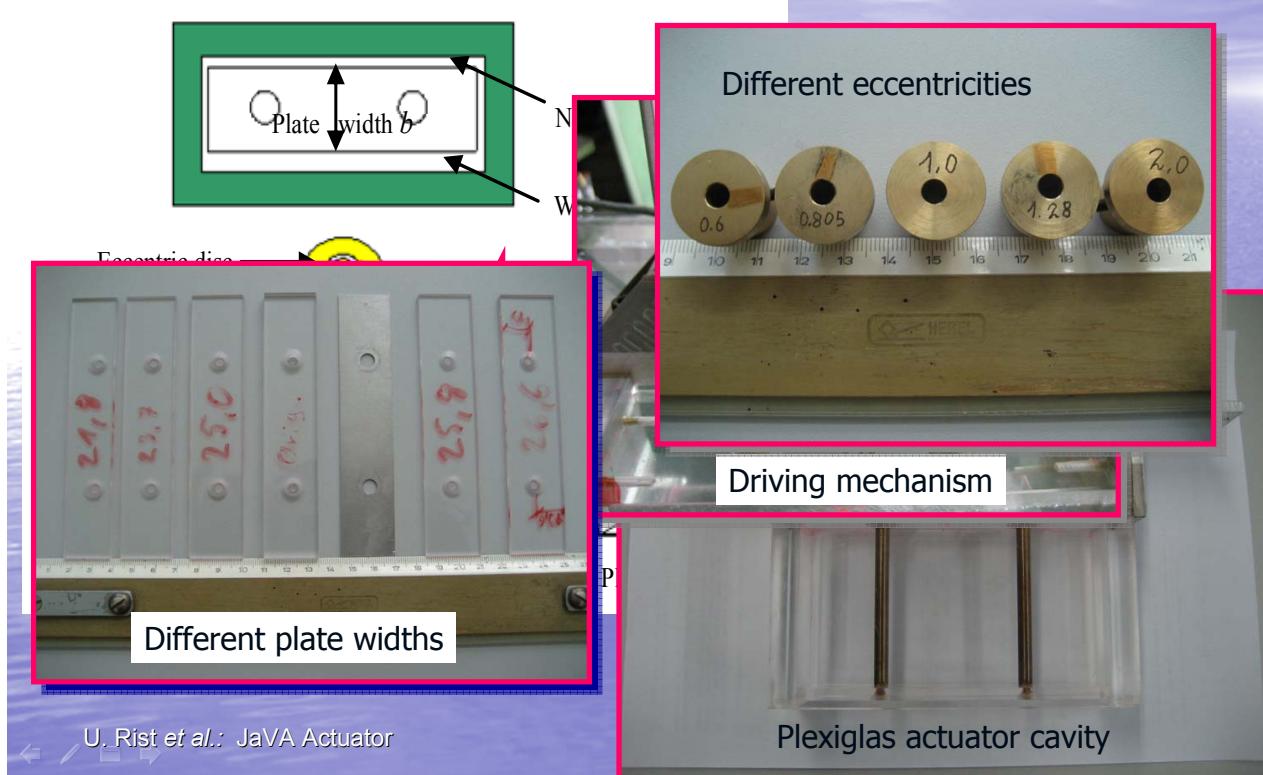
Figure 4. Horizontal flow: $f=210$ Hz, $g_r=3$, $S_a=0.11$, $Re=134$

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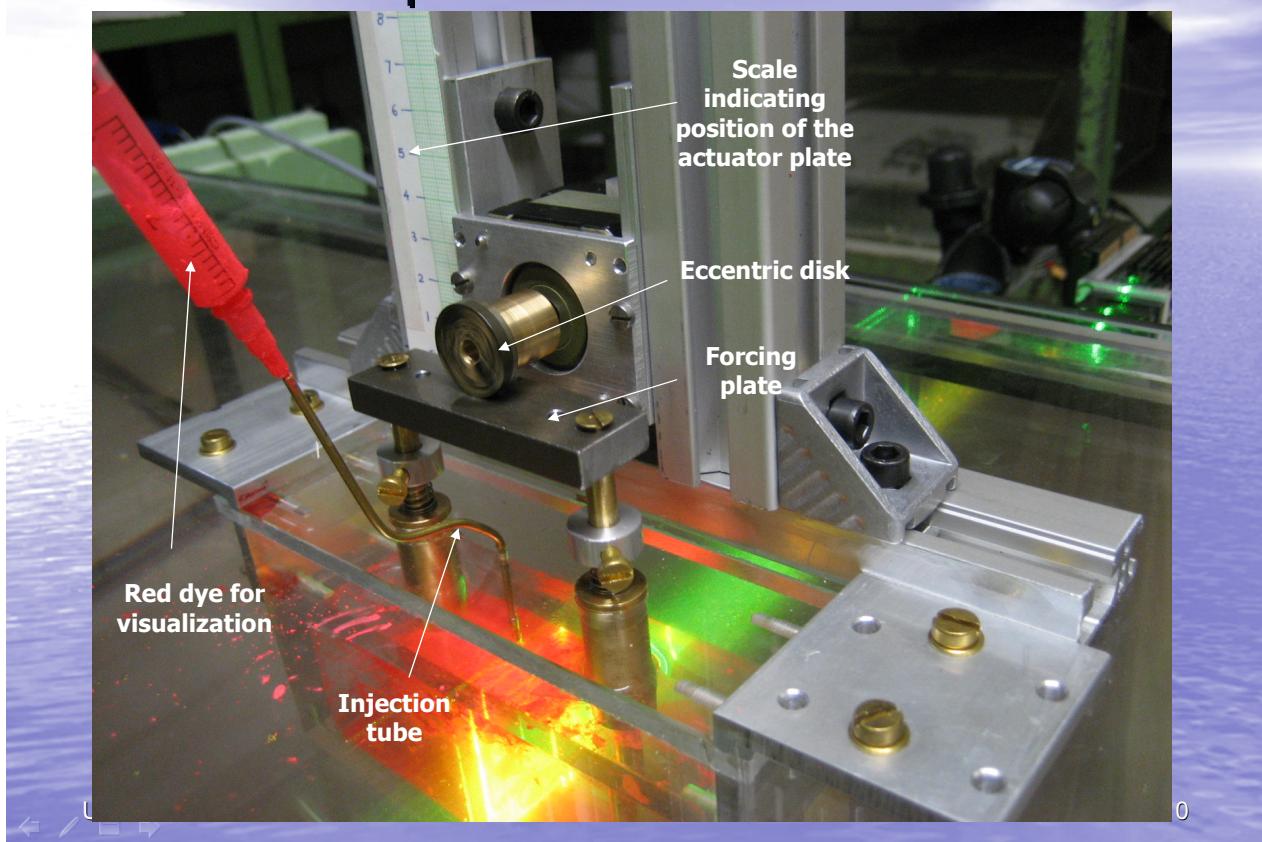
CFD Results of Rashad & Rist



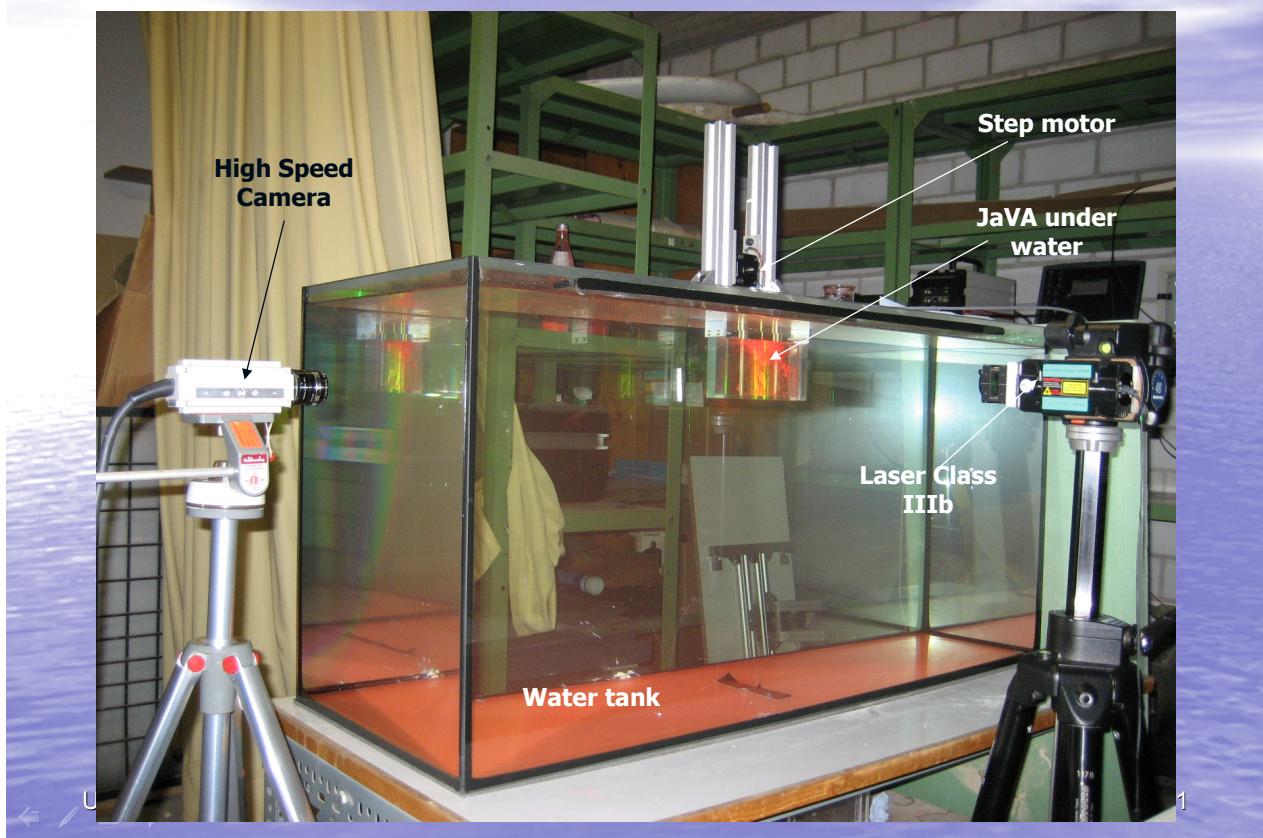
Construction of a JaVA for Experiments in Water



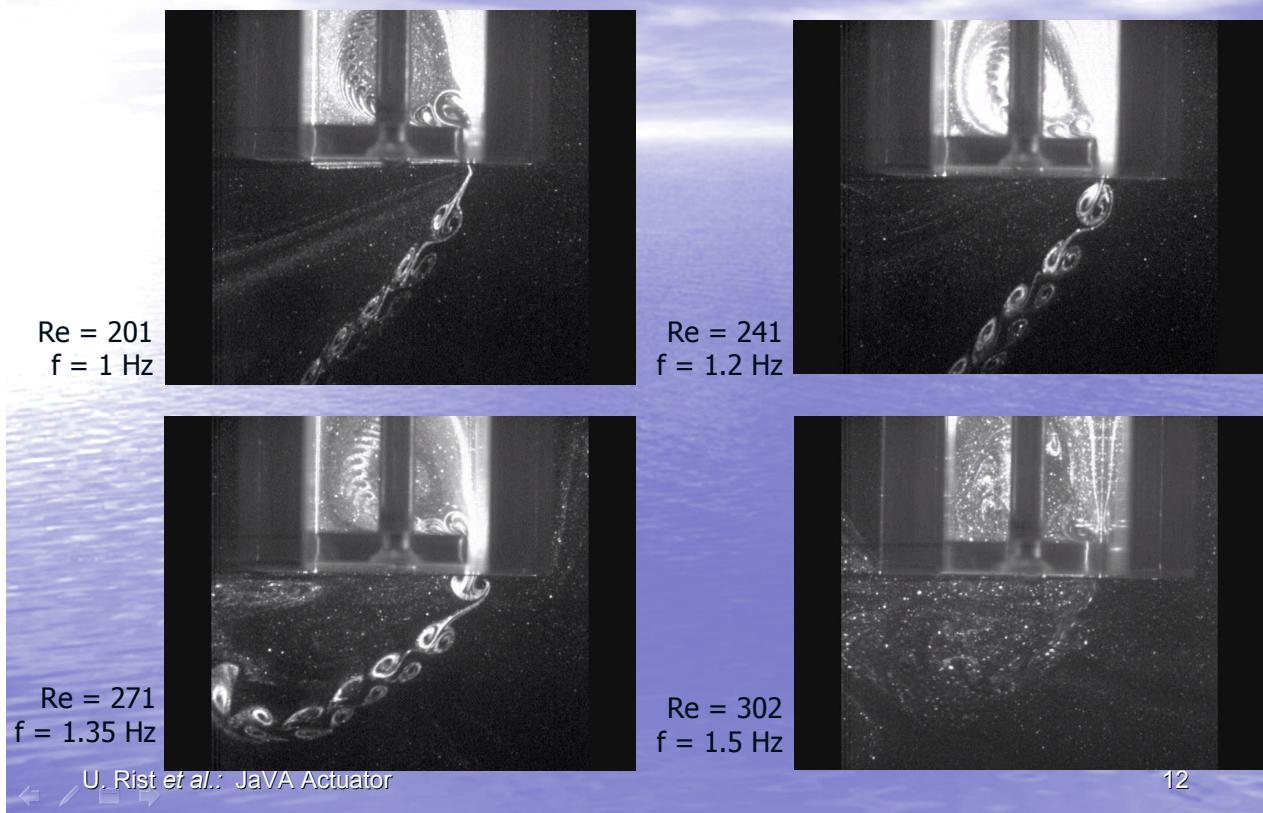
Operation of the JaVA



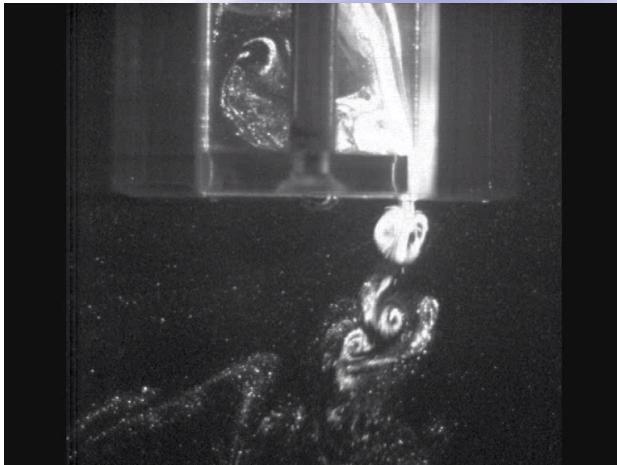
View of Experimental Setup



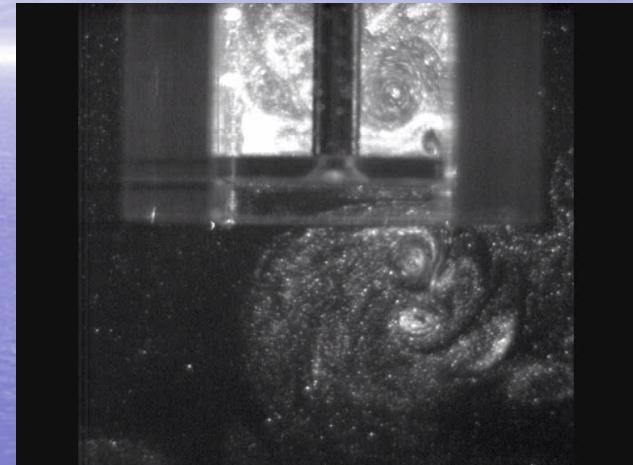
Effect of Reynolds Number



Typical JaVA-induced Unsteady and Vortex Flows



$Re = 402$, $S_a = 0.32$, $g_w = 0.1$
($a = 1.28$, $b = 25$, $f = 2$ Hz)

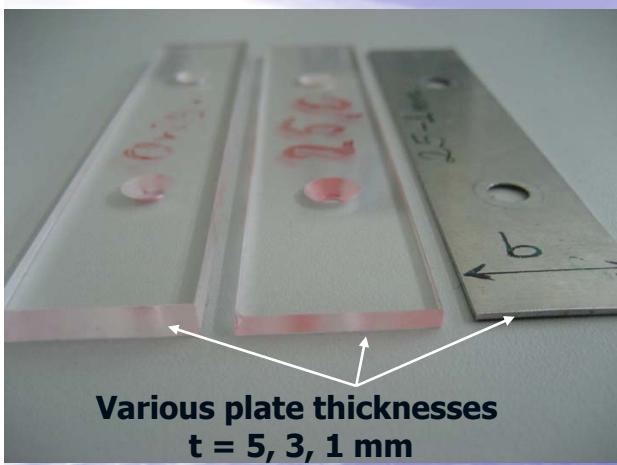


$Re = 214$, $S_a = 0.30$, $g_w = 0.034$
($a = 1.28$, $b = 26.6$, $f = 1$ Hz)

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Influence of Additional Parameters?



Various plate thicknesses
 $t = 5, 3, 1$ mm

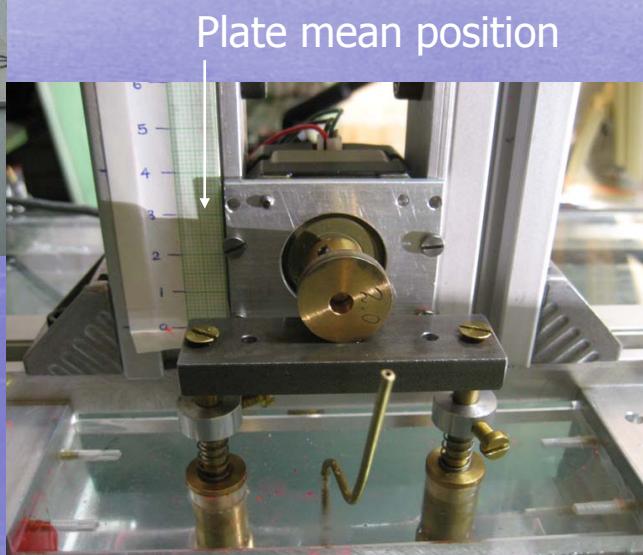


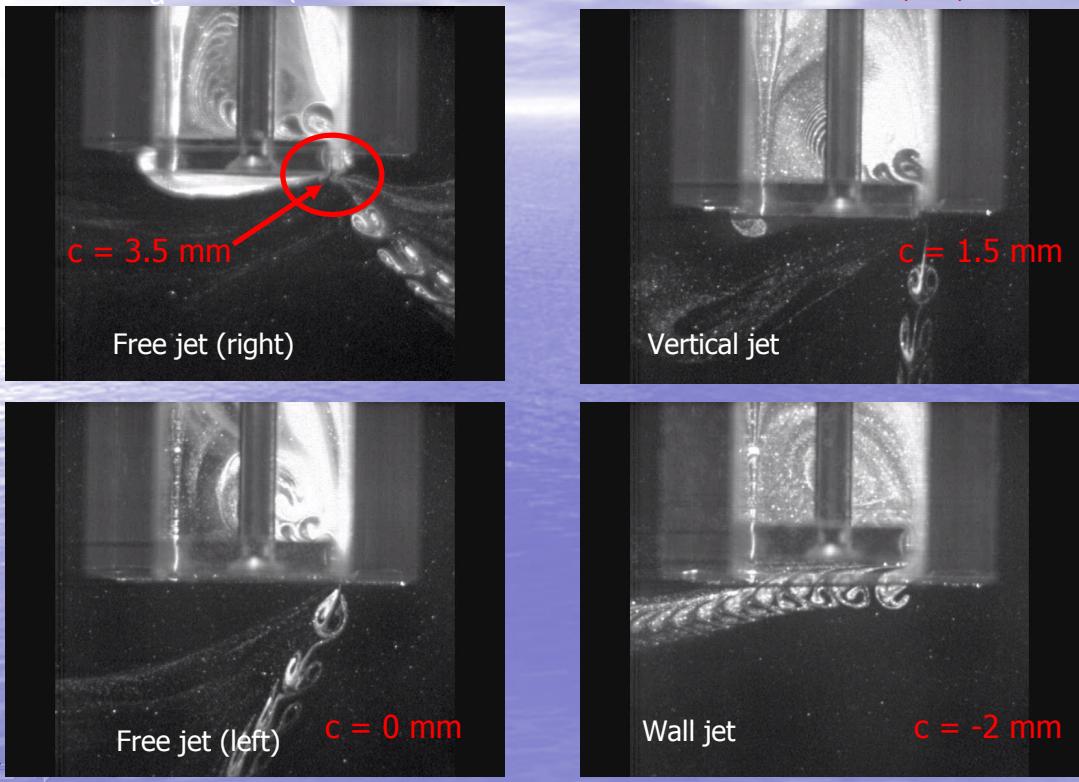
Plate mean position

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Effect of Plate Mean Position

$P = 10 \text{ Pa}, S_a = 0.32 \text{ (} a = 1.28, b = 25, f = 1 \text{ Hz, } c = -2, \dots, 3.5 \text{ mm)}$



Optical Flow Concept

Brightness change constraint. A common assumption on optical flow is that the image brightness $g(\mathbf{x}, t)$ at a point $\mathbf{x} = [x, y]^T$ at time t should only change because of motion. Thus, the total time derivative,

$$\frac{dg}{dt} = \frac{\partial g}{\partial x} \frac{dx}{dt} + \frac{\partial g}{\partial y} \frac{dy}{dt} + \frac{\partial g}{\partial t} \quad (10.1)$$

needs to equal zero. With the definitions $f_1 = dx/dt$ and $f_2 = dy/dt$, this directly yields the well-known motion constraint equation or brightness change constraint equation, BCCE [6]:

$$(\nabla g)^T \mathbf{f} + g_t = 0 \quad (10.2)$$

where $\mathbf{f} = [f_1, f_2]^T$ is the optical flow, ∇g defines the spatial gradient, and g_t denotes the partial time derivative $\partial g / \partial t$.

Optical Flow Concept

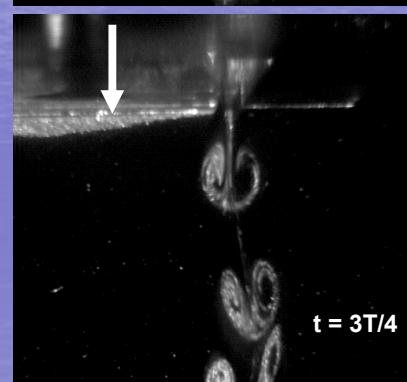
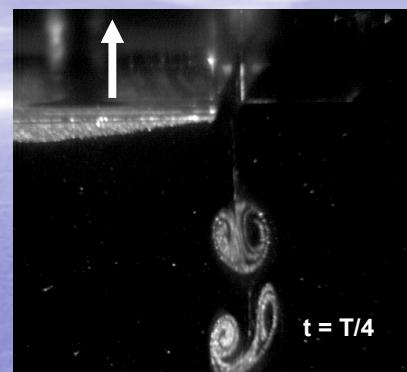
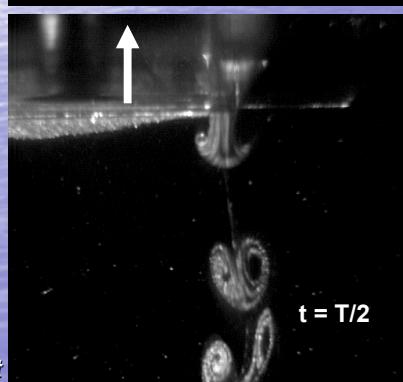
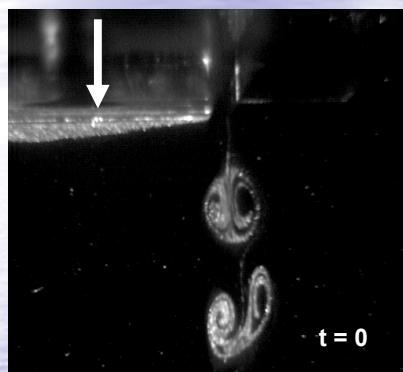
- Cannot be solved for a single point in the image, so some constraints are to be applied
- Assume that in a small neighborhood the velocity (optical flow) is constant
- Combining the pixels in the neighborhood we obtain an over determined system of equations
- Solve for $v_x = f_1$ and $v_y = f_2$ with the pseudo-inverse (based on least squares) method

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Series of Snapshots from Movie

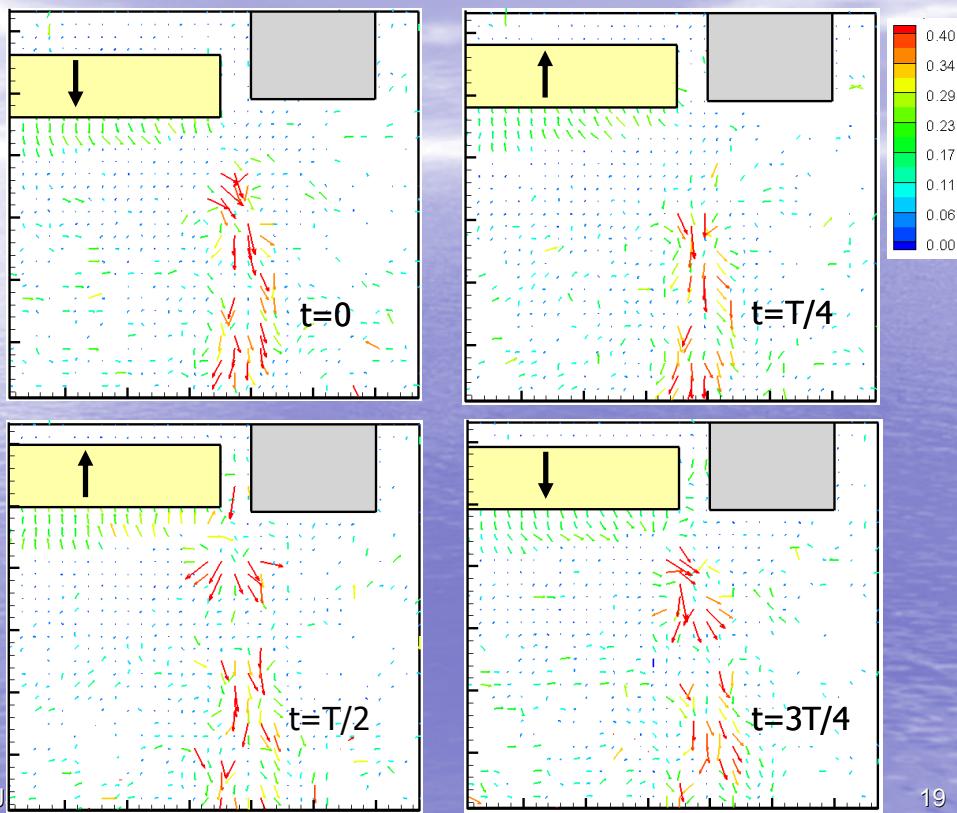
– 201, $S_a=0.32$ ($a=1.28$ mm, $b=25$ mm, $f=1$ Hz, $c=2$ mm)



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Extracted Flow Field

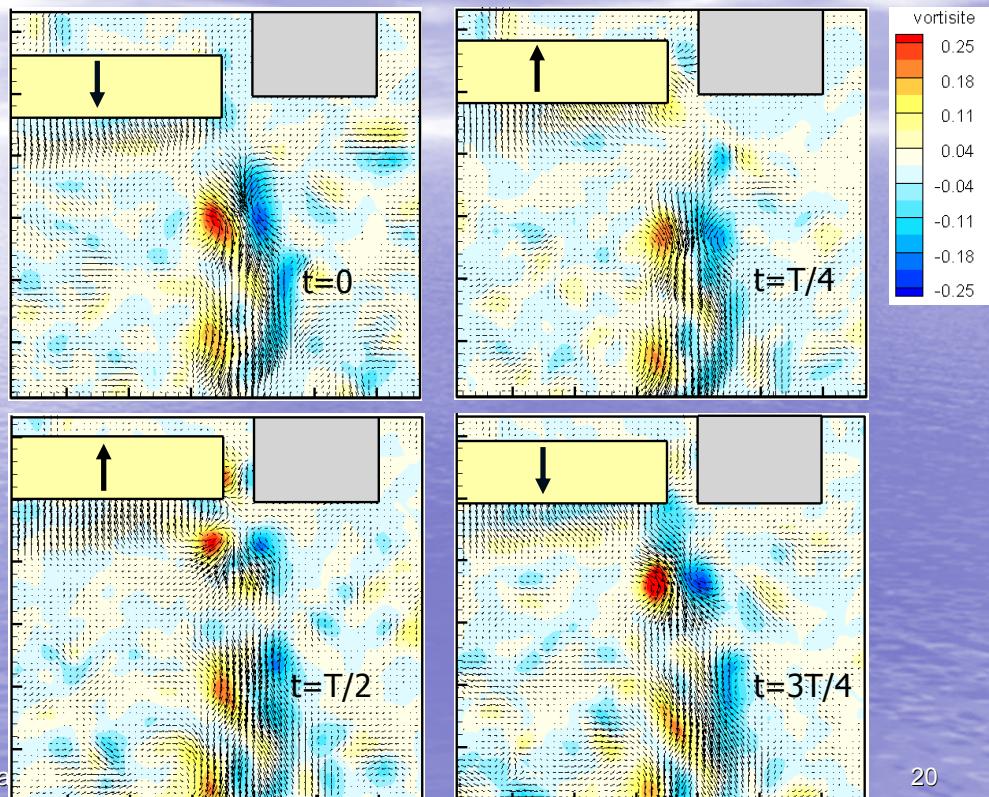


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Instantaneous Vorticity Field

Solution Enhancement and Smoothing of Optical Flow via Kriging

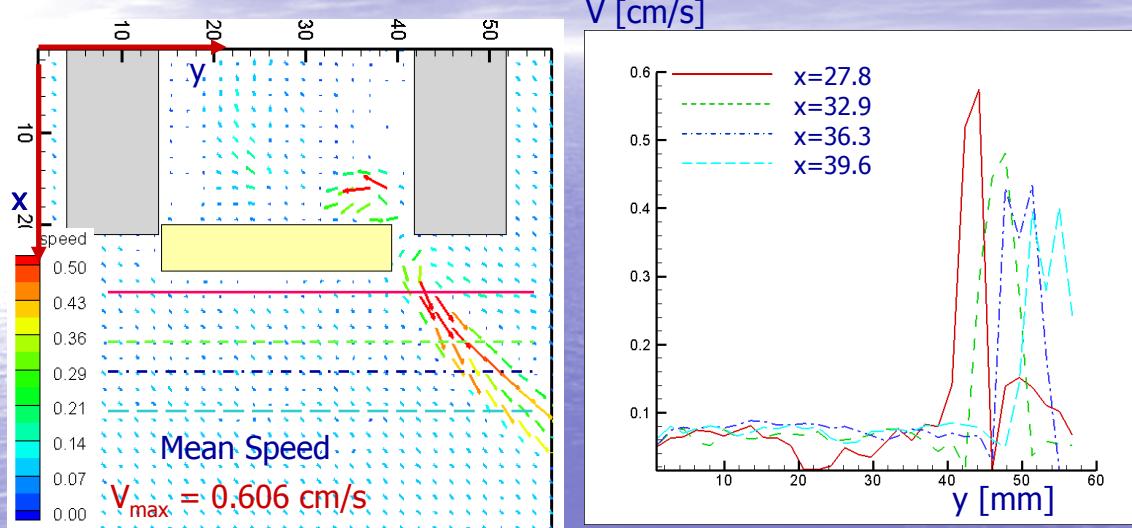


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Averaged Flow Fields (1)

$Re = 201, S_a = 0.32$ ($a = 1.28 \text{ mm} / b = 25 \text{ mm} / f = 1\text{Hz} / 3.5 \text{ mm}$)



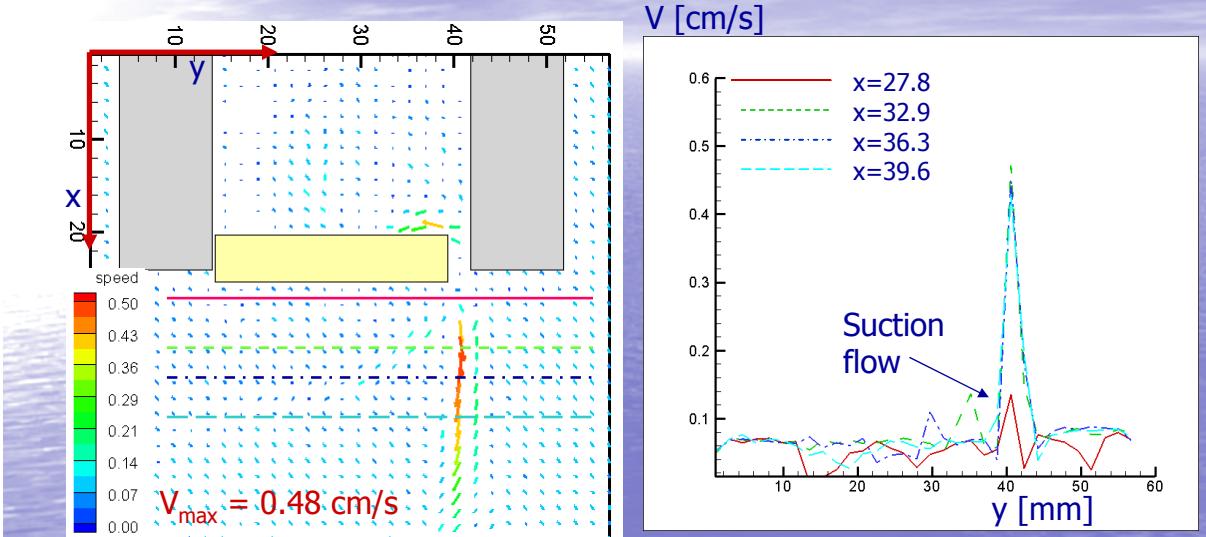
Angled Jet to the Right

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Averaged Flow Fields (2)

$Re = 201, S_a = 0.32$ ($a = 1.28 \text{ mm} / b = 25 \text{ mm} / f = 1\text{Hz} / 2 \text{ mm}$)



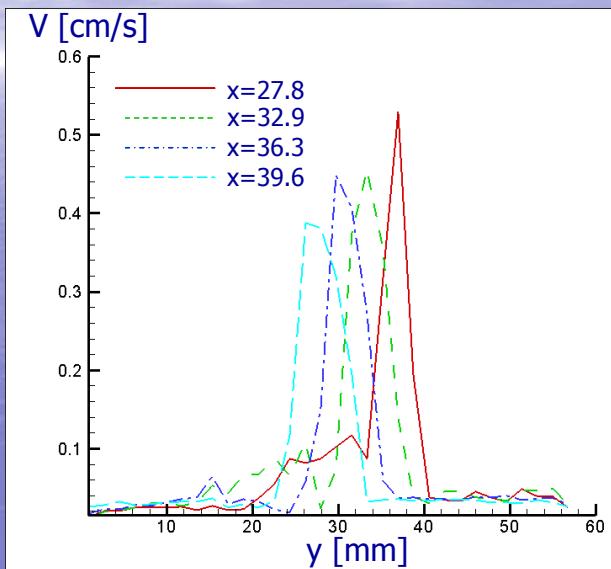
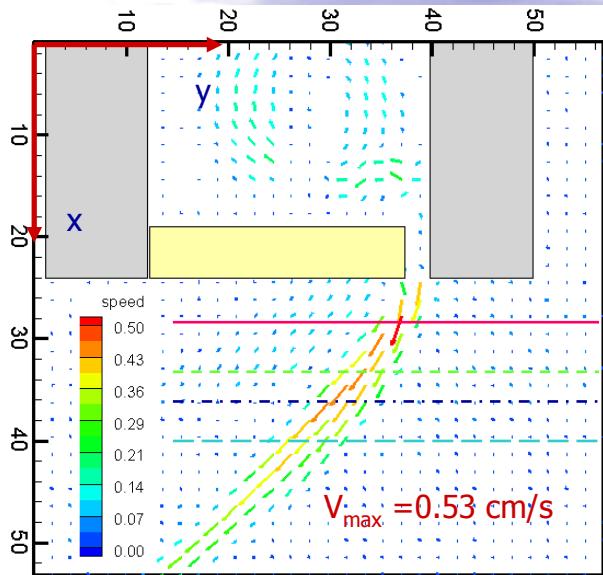
Free Vertical Jet

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Averaged Flow Fields (3)

$Re = 201, S_a = 0.32$ ($a = 1.28 \text{ mm}$ / $b = 25 \text{ mm}$ / $f = 1\text{Hz}$ / 0mm)



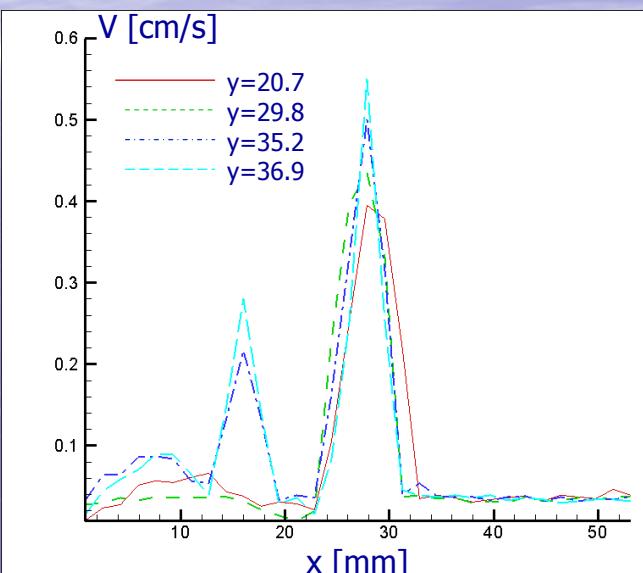
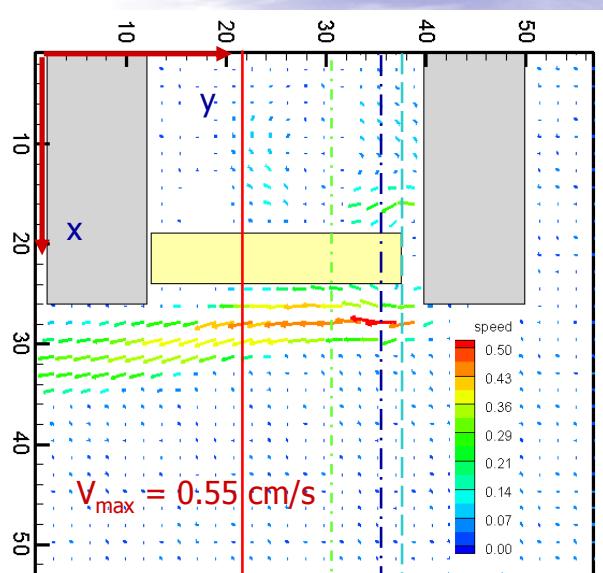
Angled Jet to the Left

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Averaged Flow Fields (4)

$Re = 201, S_a = 0.32$ ($a = 1.28 \text{ mm}$ / $b = 25 \text{ mm}$ / $f = 1\text{Hz}$ / -2 mm)



Wall Jet

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Conclusions & Outlook

- Results of Lachowicz *et al.* repeated
- Parameters not documented by Lachowicz *et al.* are of influence
- Jet flows in various directions can be obtained by changing the mean position of the actuator plate
- Accurate quantitative data can be extracted via optical flow
- The typical JaVA-induced flows are also obtained via CFD simulations which give detailed information in the close proximity of the gaps
- Kriging and constraint interpolation techniques can be used for reconstruction, smoothing and enhancement of optical flow & PIV data
- Three-dimensional effects are confined to the ends of the JaVA for most flow types
- Preliminary results of JaVA mounted in a boundary layer look very promising