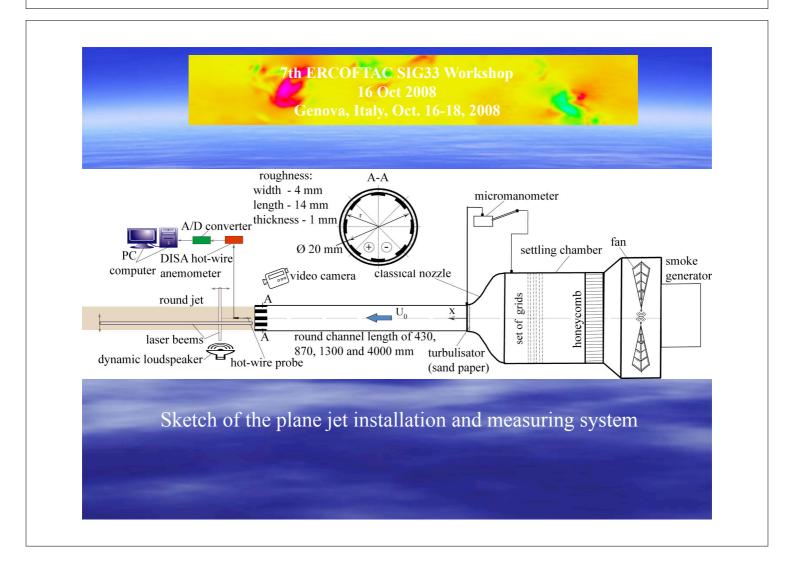
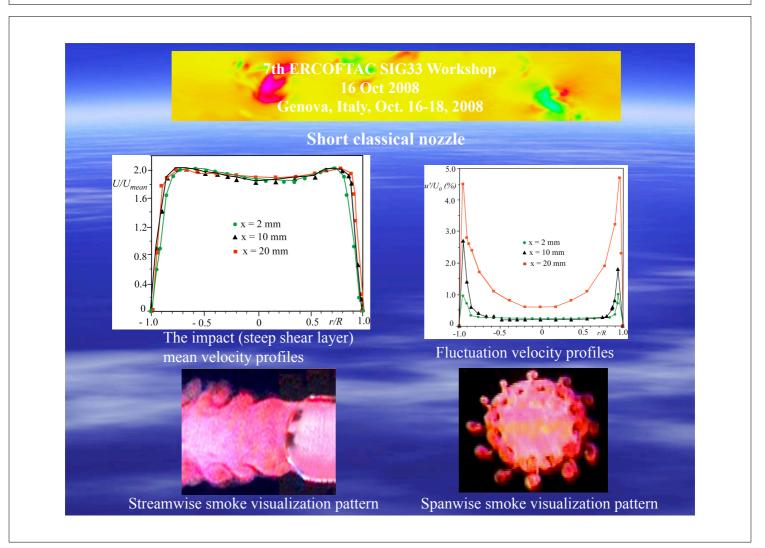
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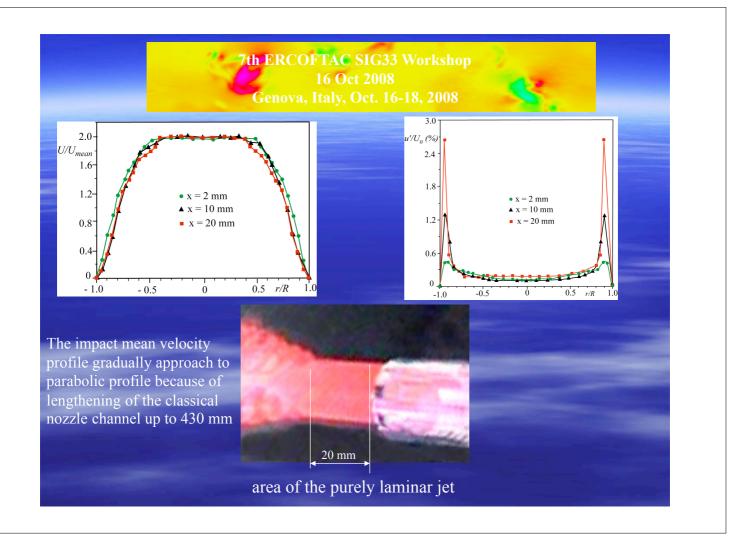
8.

Smoke visualization of the round jet evolution downstream at the different distances from a nozzle exit





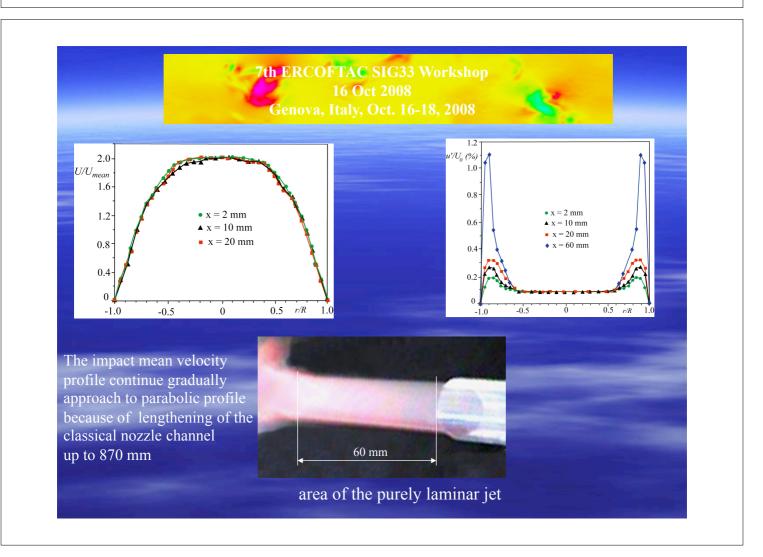




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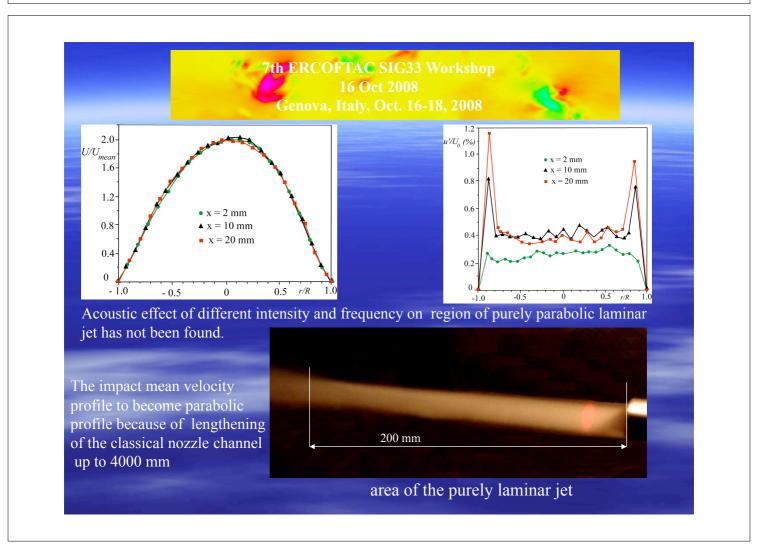
Smoke visualization video of the round jet evolution at the lengthening of the classical nozzle channel to 430 mm

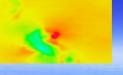




Smoke visualization video of the round jet evolution at the lengthening of the classical nozzle channel to 870 mm

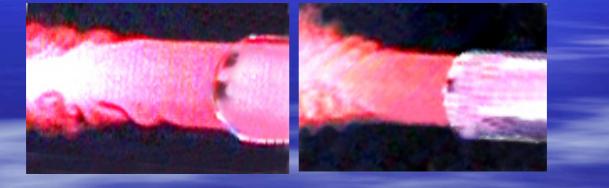




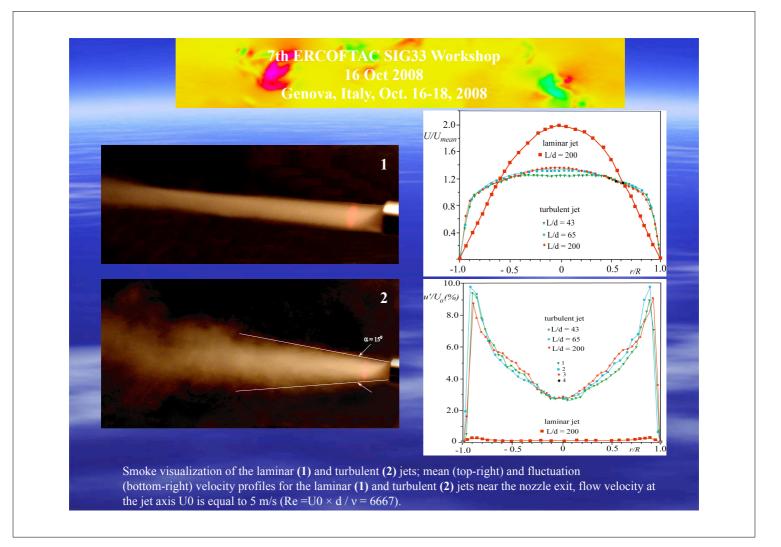


Acoustic effect on the classical round jet

An effect of acoustic forcing on the ring vortices developing in the classical laminar jet has been found. It is shown, that their scale depends on the frequency of acoustic oscillations.



Smoke visualization of the classical jet under the effect of acoustic excitation at 110 Hz (left) and 250 Hz (right), flow velocity at the jet axis U0 is equal to 5 m/s (Re = $U0 \times d / v = 6667$).



CONCLUSIONS:

• Experimental data on control of a round jet varying the initial conditions, that is, the distributions of mean and fluctuation velocity components close to the nozzle exit, are presented.

• Both laminar and turbulent regimes of the jet are considered involving results of smoke visualization of the flow patterns performed at one and the same Reynolds number.

• Variation of the initial conditions near the exit of the round jet nozzle may have a pronounced effect upon the jet structure and its evolution characteristics. In particular, transformation of the mean velocity distribution at the nozzle from the classical one with a flat section in the jet core to a parabolic profile, results in an extended puraly laminar flow region and suppression of the ring vortices.

• In the case of parabolic mean flow distribution, the puraly laminar jet is observed up to 200 mm downstream of the nozzle at the exit diameter of 20 mm. Also, such variation of the initial conditions reduces the intensity of velocity fluctuations from about u'/U0 = 1 % in the shear layer and 0.4 % in the jet core to 0.3 % in the entire cross section of the jet, where U0 is the mean velocity at the jet axis.

